

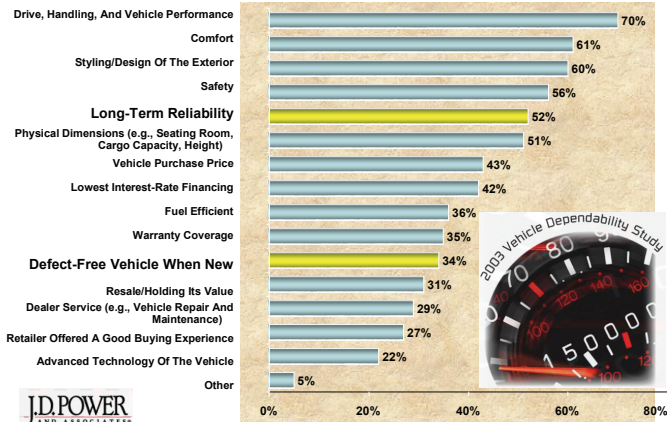
## Using LS\_DYNA to find Failure modes during design process

Dr Tayeb Zeguer  
Jaguar & Land Rover

### **Contents**

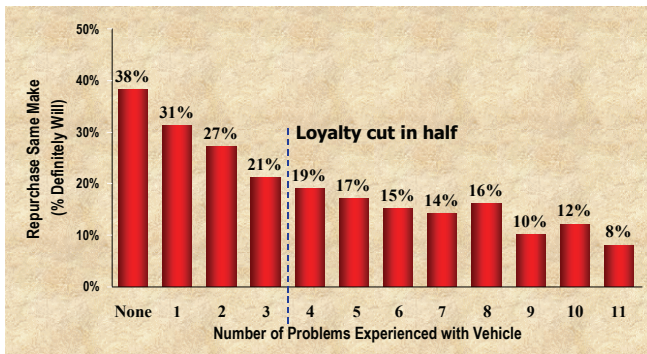
- Understanding the customer needs
- Failure Mode List Generation
- Detection using CAE – Capability and Enablers
- CAE Execution Process and the use of DFSS principles
- Conclusions

**Things demanded from us by our customers  
attributes with the most impact on purchase decision.**

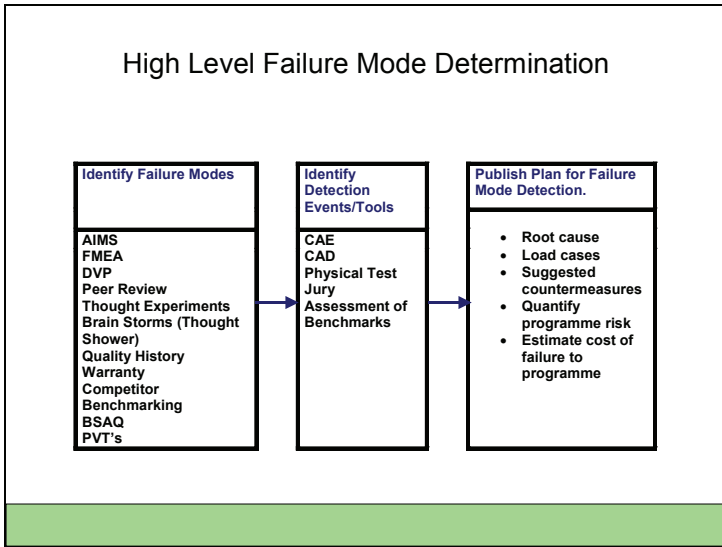
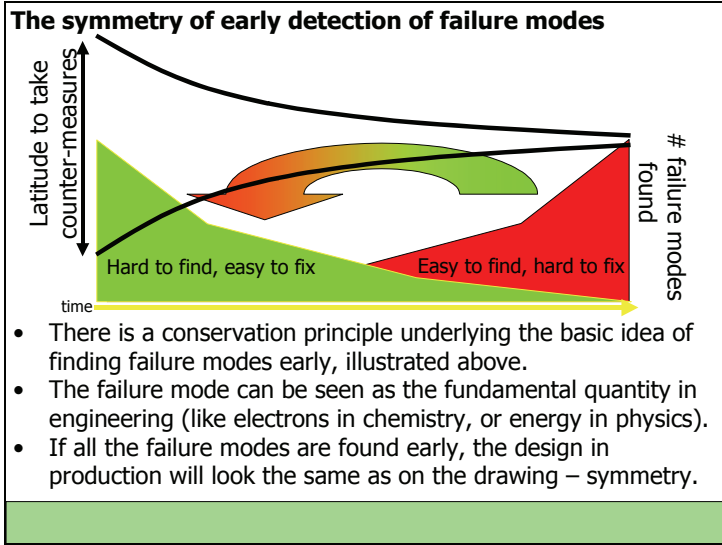


2003 Vehicle Dependability Study

**A significant drop in vehicle repurchase intent is  
due to customers finding failure modes on our behalf.**



2003 Vehicle Dependability Study

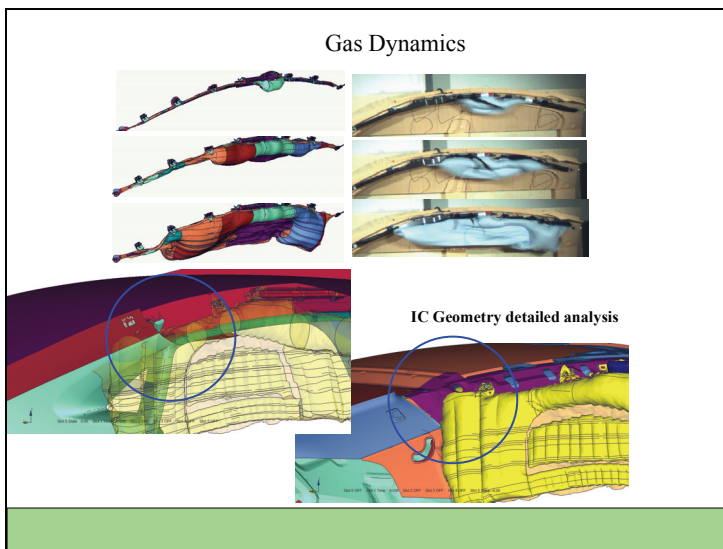
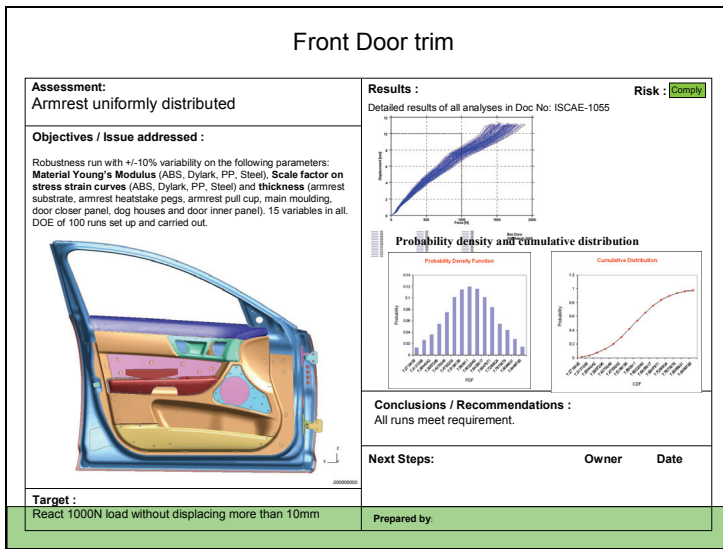


### Determination of CAE Failure Mode Activities

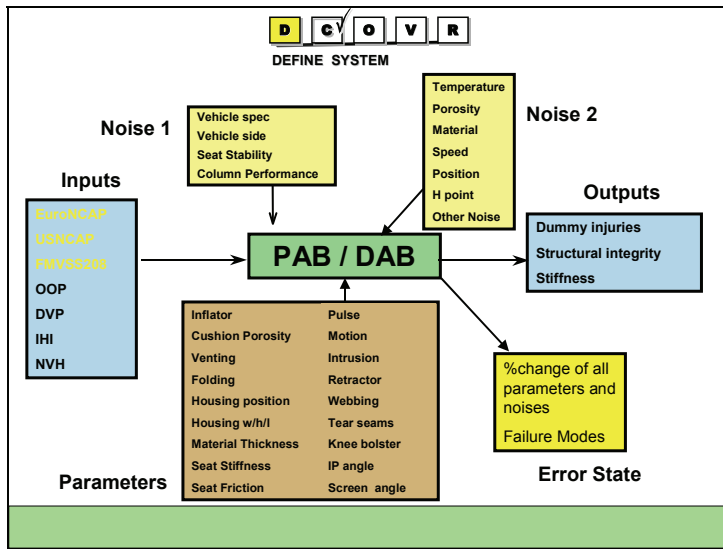
- The High Level Tracker and judgement on the failure modes that CAE can detect is carried out:
  - NOW – with a high level of confidence
  - NOW – but with some current methods development (potentially 'new' CAE activity using known codes)
  - NOW – Low level of confidence using current techniques
  - Enablers
    - What methods need developing in order to improve detection
    - What new tools need procuring and/or developing to enable detection on future programmes.

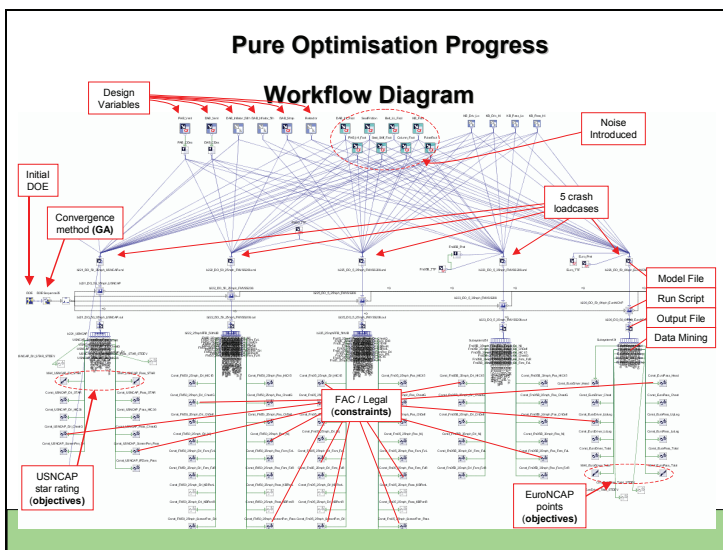
### Enablers

- CAE engineers
- Batch Meshing
- Material database
- Auto Assembly
- CAE Wrapping Process and morphing
- Auto Post
- CPUs....
- CAE PIM



## Robustness Studies Using DFSS





## Optimisation Model Setup

1. Initially a Genetic Algorithm (GA) was used to control a pure optimisation run
  - Searching over 20 generations for solutions which met all FAC / legal requirements and optimised USNCAP and EuroNCAP performance. The optimised solution was then to be used as a seed for a subsequent robustness analysis. *Low robustness was found for all optimised points*
2. GA was then used with a Multi Objective Robust Design (MORDO) option to search for the most optimum **robust** solution.
  - Rather than applying the pass/ fail criteria to injuries from individual model runs, it assesses the cloud generated by running a suite of models with noise applied around the nominal model. *15 models were run around each setup and the mean and standard deviation for each injury parameter was considered*

## Design Variable Setup

- DAB vents size 2x23mm – 2x33mm
- PAB vent size 2x44mm – 2x56 (Later increased to 2x55mm – 2x70mm)
- DAB inflator selection for 50<sup>th</sup>ile Dual stage Dt5ms v Dt10ms
- DAB inflator selection for 5<sup>th</sup>ile Single 1 v Dual stage Dt10ms
- DAB tether length 300mm – 380mm
- All airbag fire times (TTF)

### Responses – Characterisation Study

Driver HIC36  
Driver ChestG 3ms  
Driver STARS

Passenger HIC36  
Passenger ChestG 3ms  
Passenger STARS

USNCAP

25mph Driver HIC15  
25mph Driver ChestG 3ms  
25mph Driver Nij  
25mph Driver Neck FzT  
25mph Driver Femur FzL  
25mph Driver Femur FzR  
25mph Driver Ch Deflection

FMVSS208  
50<sup>th</sup> ile UB  
25mph  
5<sup>th</sup> ile UB  
25mph  
5<sup>th</sup> ile Belted  
35mph

Driver Head  
Driver Chest  
Driver Upper leg  
Driver Lower leg  
Driver Total Frontal Points

Euro NCAP

25mph Passenger HIC15  
25mph Passenger ChestG 3ms  
25mph Passenger Nij  
25mph Passenger Neck FzT  
25mph Passenger Femur FzL  
25mph Passenger Femur FzR  
25mph Passenger Ch Deflection

Passenger Head  
Passenger Chest  
Passenger Upper leg  
Passenger Lower leg  
Passenger Total Frontal Points



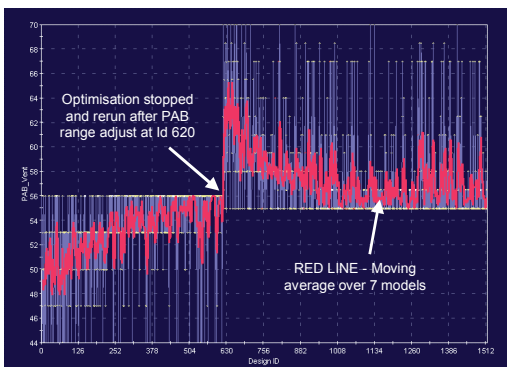
## History Chart

1. A useful way to chart the progress and trends of input / output variables over generations (design Id)
  - The chart shows the trends throughout the genetic search and allows the user to see the direction it is taking
2. Shows if convergence of an input variable is occurring towards a value as the analysis progresses
  - If this value is also the limit of the range then the process may be restarted with modifications to the input ranges (PAB vent size in next slide)
3. Shows if convergence of an output variable is occurring towards a particular value
4. A moving average is used to emphasise the trends within the genetic search for solutions

## Pure Optimisation Progress

### PAB Vent size convergence

PAB vent input range was altered at Run Id 620 due to convergence towards a limit



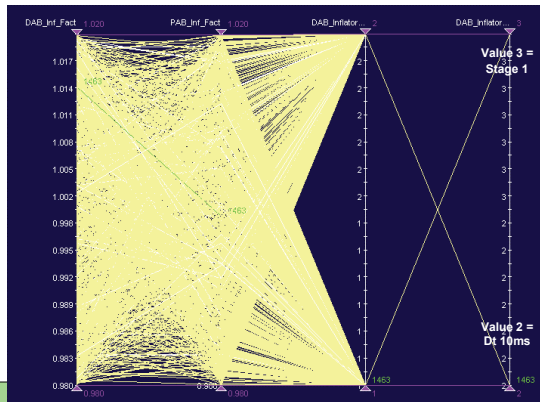
## Parallel Chart

1. A very useful way to understand relationships between input-input, input-output and output-output in the data table
2. Each axes represent an input/ output variable, showing the full valid range it could take
3. Coloured lines represent models; plotted across the various axis at the particular values for that model
4. Dynamic filtering by input/output value is used to retain models which fulfil certain criteria
5. Models (coloured lines) can also be turned on or off according to rules such as model status (feasible, unfeasible, error, marked, group, etc)

6. **By removing unfeasible (yellow) and error (red) models, it is possible to clearly see the range of inputs which leads to feasible (white) models**

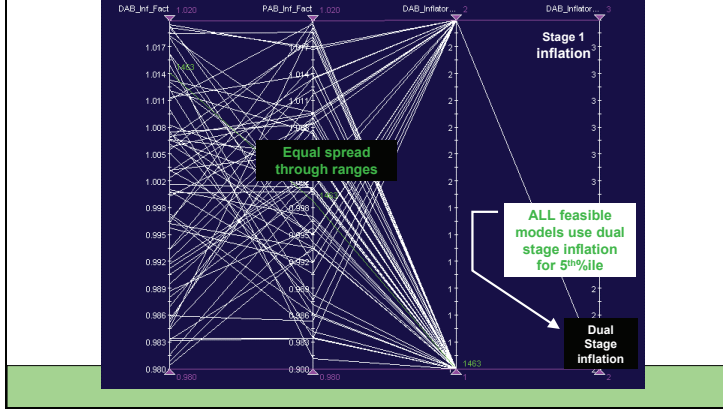
## DAB inflator selection for 5<sup>th</sup> %ile

All models shown – very little can be concluded as coloured lines are equally distributed through input combinations



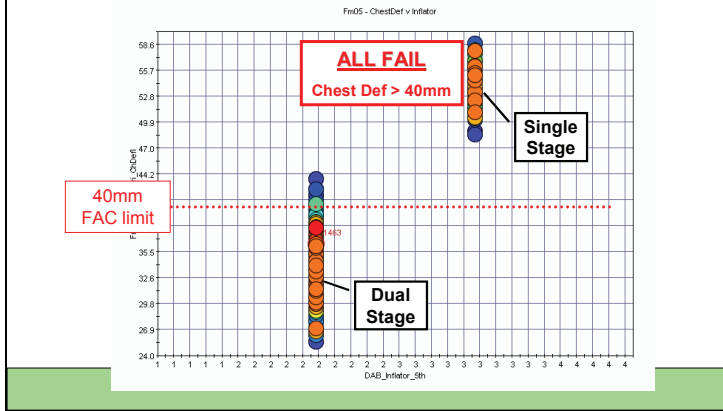
### Only feasible models shown

The system is insensitive to inflator power variations and 50<sup>th</sup> DAB inflator as shown by the even spread of models though the input parameters. ALL lines go through the lower point for the 5<sup>th</sup>ile DAB inflator representing a clear Dt10ms choice for ALL feasible models



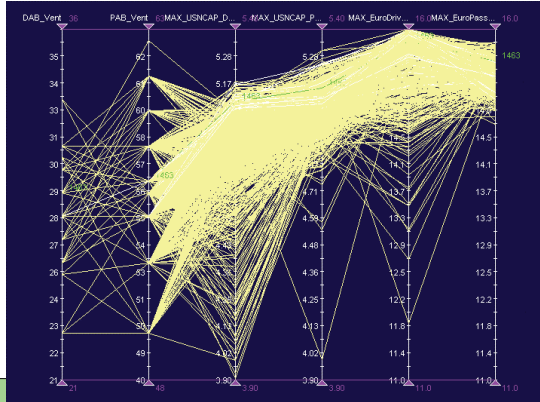
### Pure Optimisation Progress

5th%ile UB Chest Deflection  
No feasible models with a single stage inflator



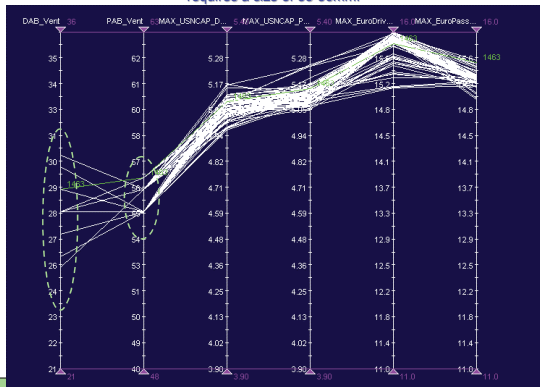
### Airbag vent sizes

All models shown – very little can be concluded as coloured lines are equally distributed through input combinations



### Only feasible models now shown

These all meet legal and FAC requirements and obtain highest NCAP ratings.  
It is clear that the valid vents for the DAB lie in the range 26-30mm whilst the larger PAB vent specifically requires a size of 55-56mm.



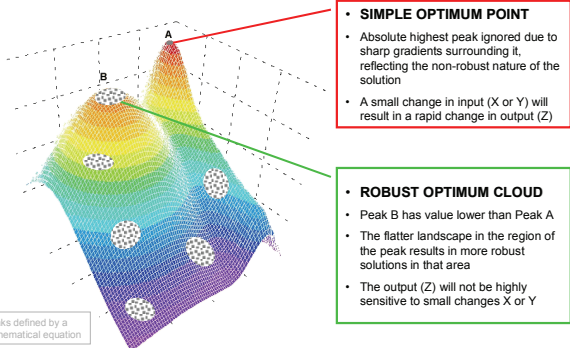
## MORDO

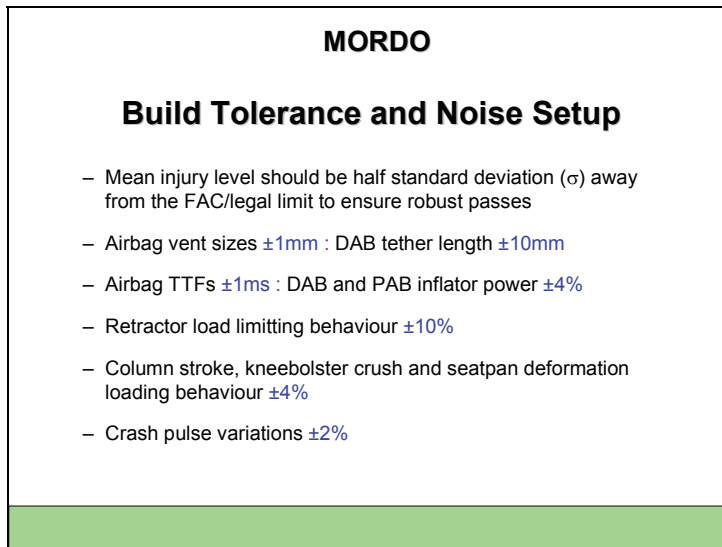
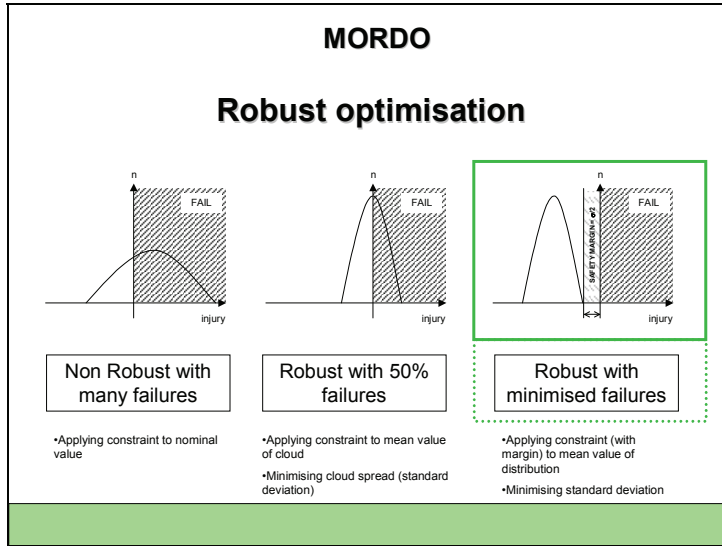
### Multi Objective Robust Design Optimisation

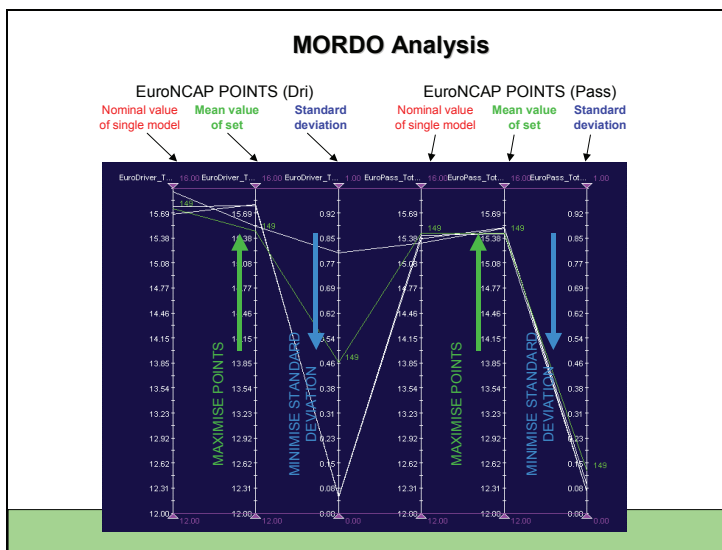
1. This functionality is provided by modeFRONTIER and is used to find the optimum of all **robust** solutions
2. Rather than running a conventional optimisation and following with a subsequent robustness run, MORDO optimises only within the solutions marked as robust (based on standard deviation of a set)
3. Robustness is defined by the standard deviation of the set around a nominal model (vent cutting tolerances, pulse severity, TTF variation, inflator output, seat stiffness and friction, etc)
4. Further generations are created to improve the **mean** value of the sets rather than absolute value of any one model

## GA using MORDO

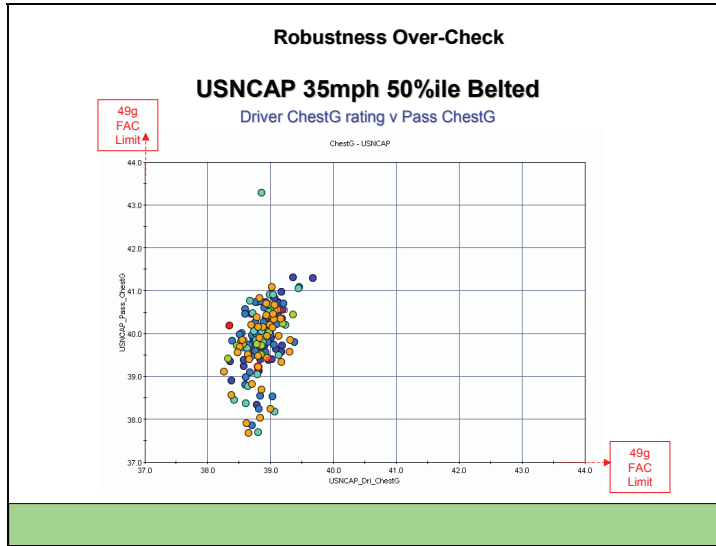
Submits sets of 'noise' models around the nominal model  
Assesses the **SPREAD** and **MEAN** value of each model cloud







- ### Robustness
- After Multi Objective Robust Design Optimisation
1. The optimised robust solution selected through the MORDO analysis is now put through a thorough robustness run
  2. An aggressive distribution is applied to the models by using UNIFORM distributions for each input variable as this maximises the number of models with variables at the extremes of the tolerance ranges. Other distributions (normal, Cauchy, etc) would create a suite of models with a bias towards the nominal value
  3. Many models (>200) are run with distribution around the nominal point.
  4. Of these runs, the number of models which pass all requirements (feasible) and the number of models which fail any (unfeasible) are used to create a 'pass/fail ratio'
  5. The 'pass/fail' ratio is a simple measure of robustness. A high ratio is desirable as this indicates a robust setup



## Conclusions

- Adding noise factors simultaneously during optimisation is the best way in producing a robust design
- If all the failure modes are found early, the design in production will look the same as on the drawing – symmetry
- LS-DYNA is a key enabler in finding most failure modes and eliminated them early in a vehicle program.
- Need to develop LS\_DYNA capabilities further to address failure modes that can't currently be modelled.