

# Preliminary Results for an Isogeometric Shell

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Research supported by NSF grant 0700204  
Additional support for Bazilevs and Hughes from ONR



# Isogeometric Analysis

- Isogeometric analysis uses NURBS as basis functions.
- NURBS are the basis functions used in CAD programs.
- Therefore: facilitates direct CAD to analysis interface.
- NURBS are nicely behaved.
  - Improved numerical conditioning.
  - Larger time step size for higher order elements than for Lagrangian polynomials.



# B-Spline Basis Functions

- Piecewise polynomials in space.
- Degree determined by the knot vector:

$$\Xi = \{\xi_1, \dots, \xi_{n+p+1}\}$$

- Coefficients of polynomials are points in space, referred to as *control points*,  $B_i$
- Basis functions are generated recursively using the knot vector starting at  $p=0$  (piecewise constants).

$$N_{i,0}(\xi) = \begin{cases} 1 & \text{if } \xi_i \leq \xi < \xi_{i+1} \\ 0 & \text{otherwise.} \end{cases}$$

$$N_{i,p}(\xi) = \frac{\xi - \xi_i}{\xi_{i+p} - \xi_i} N_{i,p-1}(\xi) + \frac{\xi_{i+p+1} - \xi}{\xi_{i+p+1} - \xi_{i+1}} N_{i+1,p-1}(\xi).$$



# Properties of B-Splines

- B-splines sum to 1 like Lagrange interpolation functions.

$$\sum_{i=1}^n N_{i,p}(\xi) = 1 \quad \forall \xi$$

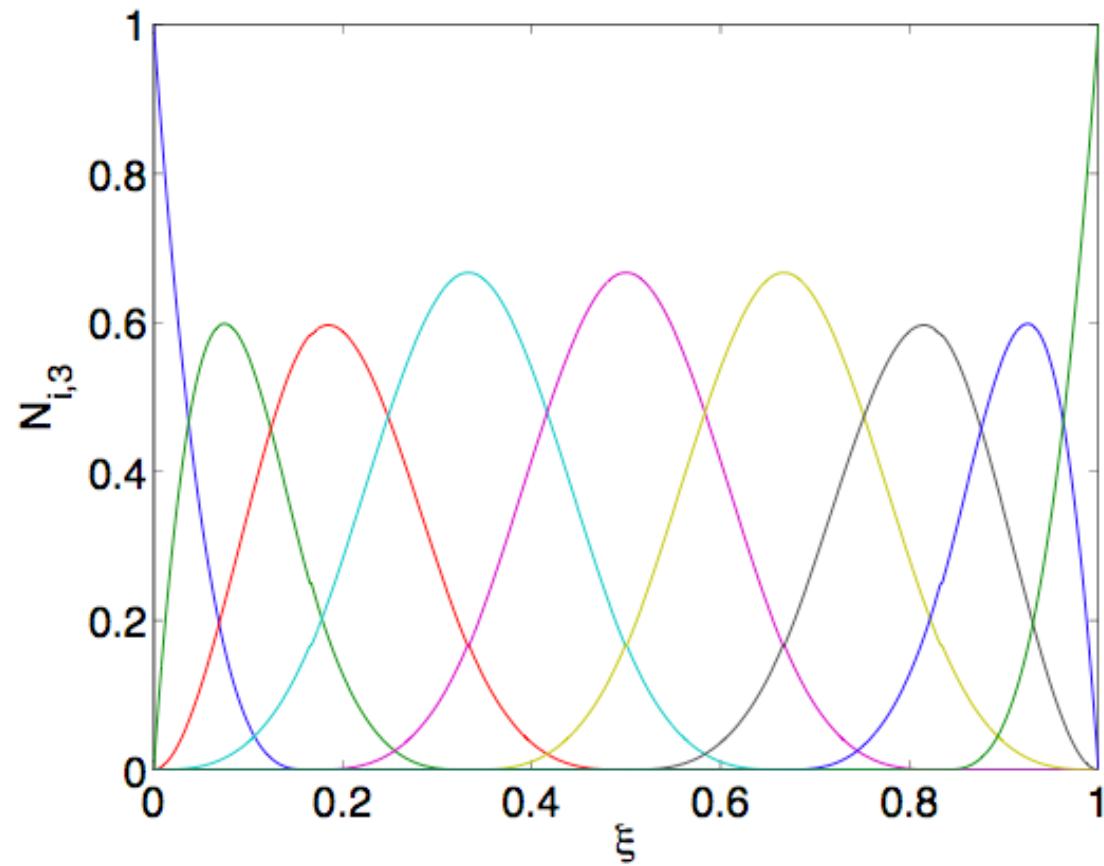
- The support of each  $N_{i,p}(\xi)$  compact and contained in the interval  $[\xi_i, \xi_{i+p+1}]$  similar to Lagrange interpolation polynomials.
- B-spline basis functions are non-negative:

(in contrast to  $L_i(\xi) \geq 0 \forall \xi$  Lagrange polynomials).



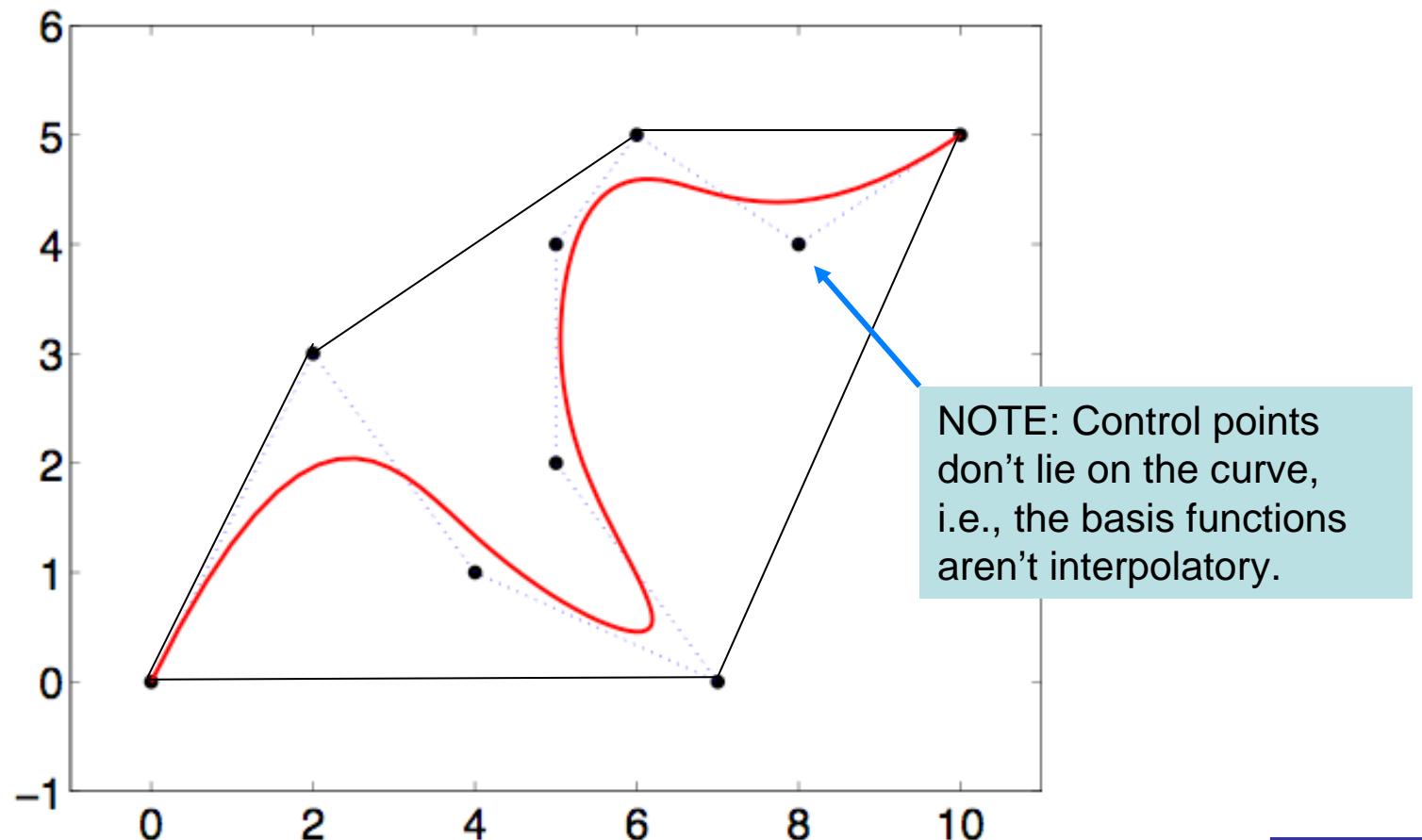
# Cubic B-Spline Basis Functions

$$\Xi = \{0, 0, 0, 0, 1/6, 1/3, 1/2, 2/3, 5/6, 1, 1, 1, 1\}$$



# Positivity Limits Oscillations

Curve lies inside the convex hull of the control points.



$$\mathbf{C}(\xi) = \sum_{i=1}^n N_{i,p}(\xi) \mathbf{B}_i$$



# B-Spline Surfaces and Solids

- Surfaces and solids are described in terms of tensor products of one-dimensional basis functions as is standard with Lagrange interpolation functions in standard FEA.

$$\mathbf{S}(\xi, \eta) = \sum_{i=1}^n \sum_{j=1}^m N_{i,p}(\xi) M_{j,q}(\eta) \mathbf{B}_{i,j} \quad \text{Surface}$$

$$\mathbf{S}(\xi, \eta, \zeta) = \sum_{i=1}^n \sum_{j=1}^m \sum_{k=1}^l N_{i,p}(\xi) M_{j,q}(\eta) L_{k,l}(\zeta) \mathbf{B}_{i,j,k} \quad \text{Solid}$$



# NURBS

- Non-Uniform Rational B-Splines (NURBS)
- Control points are homogenous coordinates.

$$(\mathbf{B}_i)_j = \frac{(\mathbf{B}_i^w)_j}{w_i}, \quad j = 1, \dots, d \quad w_i = \text{weights}$$

- Basis functions:  $R_i^p(\xi) = \frac{N_{i,p}(\xi)w_i}{\sum_{\hat{i}=1}^n N_{\hat{i},p}(\xi)w_{\hat{i}}}$
- Curve:  $\mathbf{C}(\xi) = \sum_{i=1}^n R_i^p(\xi) \mathbf{B}_i$



# NURBS Properties

- NURBS basis functions sum to 1.

$$\sum_{i=1}^n R_i^p(\xi) = 1 \quad \forall \xi$$

- Continuity and support are the same as B-Splines.
- If all weights are equal, NURBS become B-Splines.
- NURBS surfaces and solids are tensor products.
- Basis functions are specified in terms of a knot vector like B-Splines.



# Isogeometric Analysis

- A mesh for a NURBS patch is defined by the product of knot vectors.
- Knot spans subdivide the domain into “elements.”
- The support of each basis function consists of a small number of elements.
- The control points associated with the basis functions define the geometry.
- The isoparametric concept is invoked, that is, the unknown variables are represented in terms of the basis functions which define the geometry.



# Shell Formulations

- 3 types currently available.
- IFORM=0: Degenerated solid element with rotational DOF.

$$v_i(\xi) = \sum_{A=1}^n N_A(\xi) \left( v_{Ai} + \frac{h\xi_3}{2} e_{ijk} \omega_{Aj} n_{Ak} \right)$$

- IFORM=2: Thin shell without rotational DOF.

$$v_i(\xi) = \sum_{A=1}^n N_A(\xi) v_{Ai} + \frac{h\xi_3}{2} \sum_{B,k} \frac{\partial n_i(\xi)}{\partial x_{Bk}} v_{Bk}$$

- IFORM=3: Reissner-Mindlin with rotational DOF.

$$v_i(\xi) = \sum_{A=1}^n N_A(\xi) \left( v_{Ai} + \frac{h\xi_3}{2} e_{ijk} \omega_{Aj} n_k(\xi) \right)$$

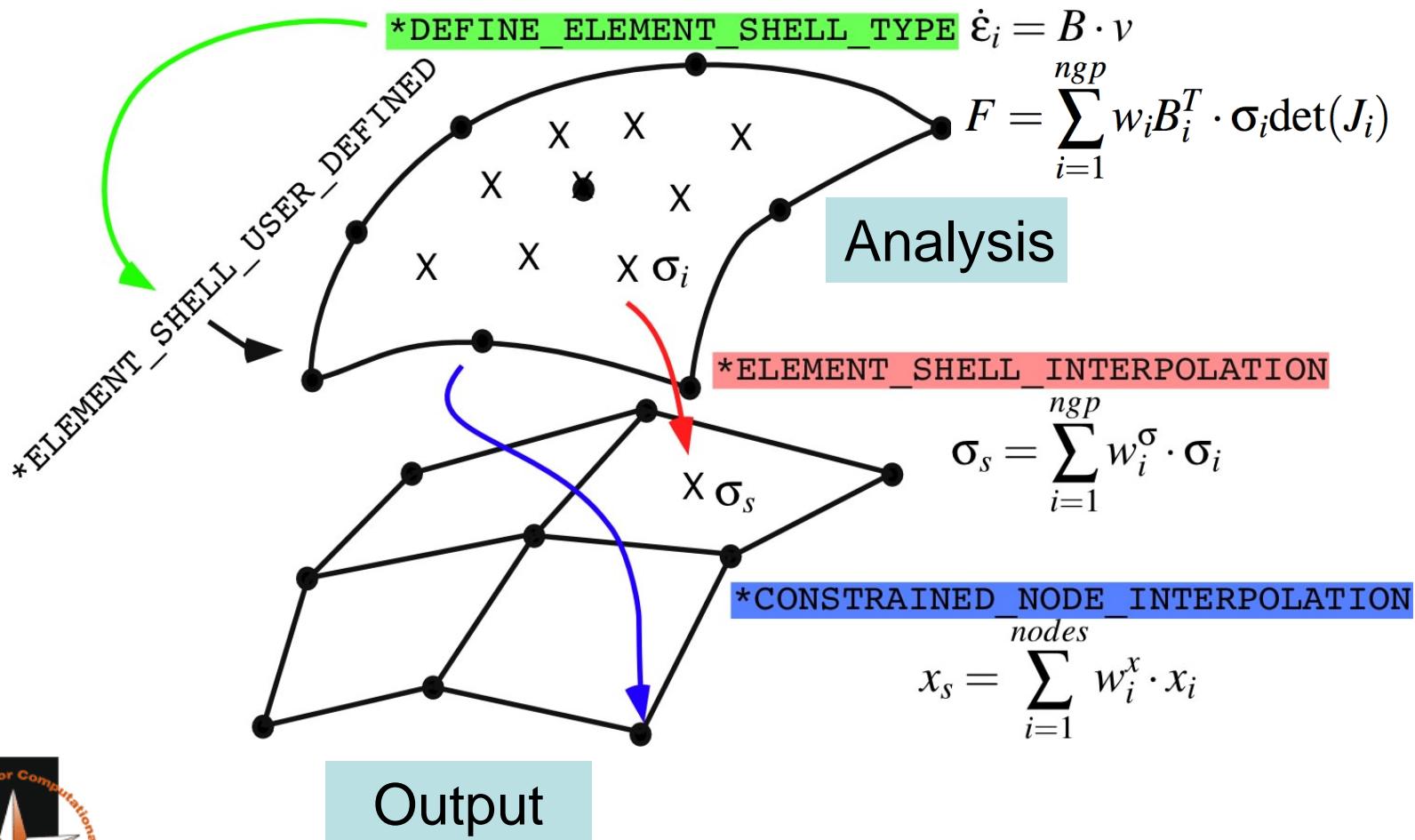


# Implementation in LS-DYNA

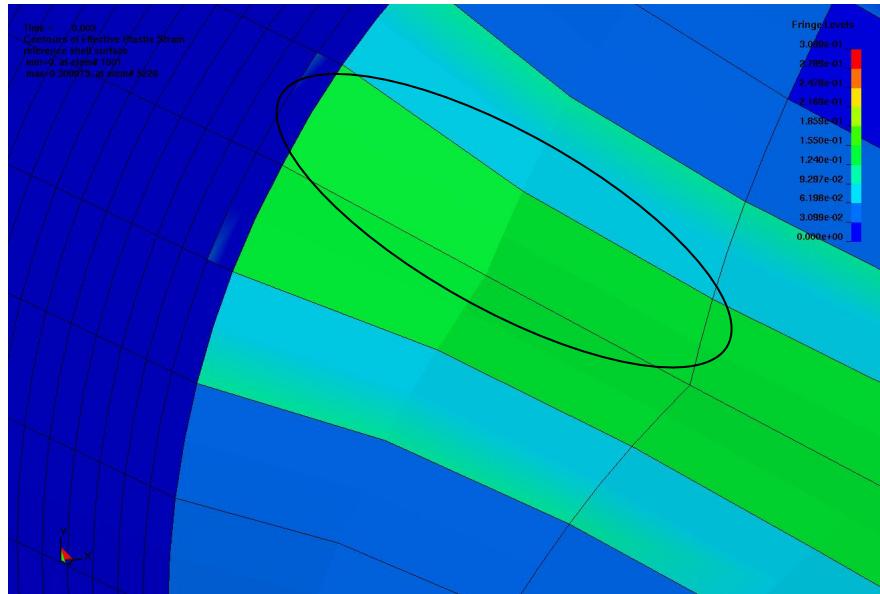
- User-defined element capability:
  - Fast prototyping of elements without programming.
  - Available for both solids and shells.
  - Executes slower than production elements.
- Some boundary conditions implemented via interpolation elements.
  - Contact doesn't have underlying smoothness of NURBS.
  - Pressure distribution is not exactly integrated.



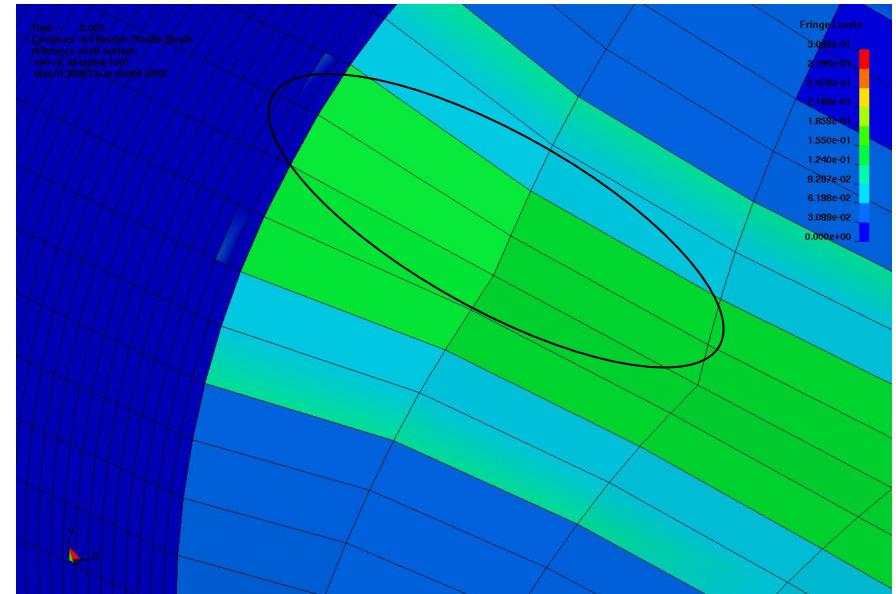
# User-Defined Elements



# User-defined Elements



User-defined element



Four interpolation elements

Later results display interpolation elements to better show the deformed geometry.



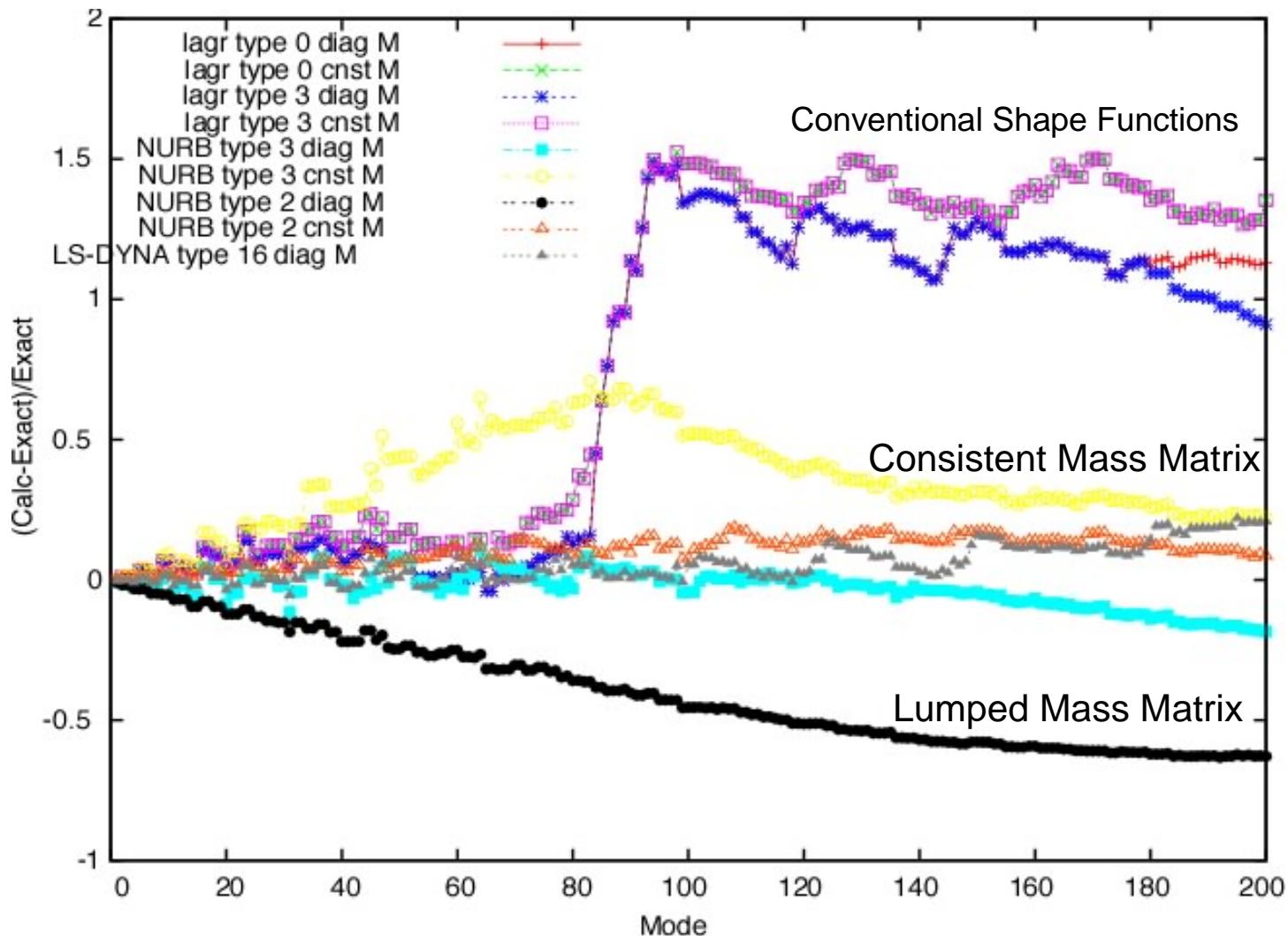
# Transverse Vibrations of a Poisson-Kirchhoff Plate

- Square plate, simply supported.
  - $[0,1] \times [0,1]$
  - Assumes unit flexural stiffness & density.
  - Exact solution:  $\omega_{mn} = \pi^2(m^2 + n^2)$
- J. A. Cottrell et al., *CMAME* 195 (2006) 5257-5296:
  - Reference for analytical solution.
  - Solution with isogeometric solids.

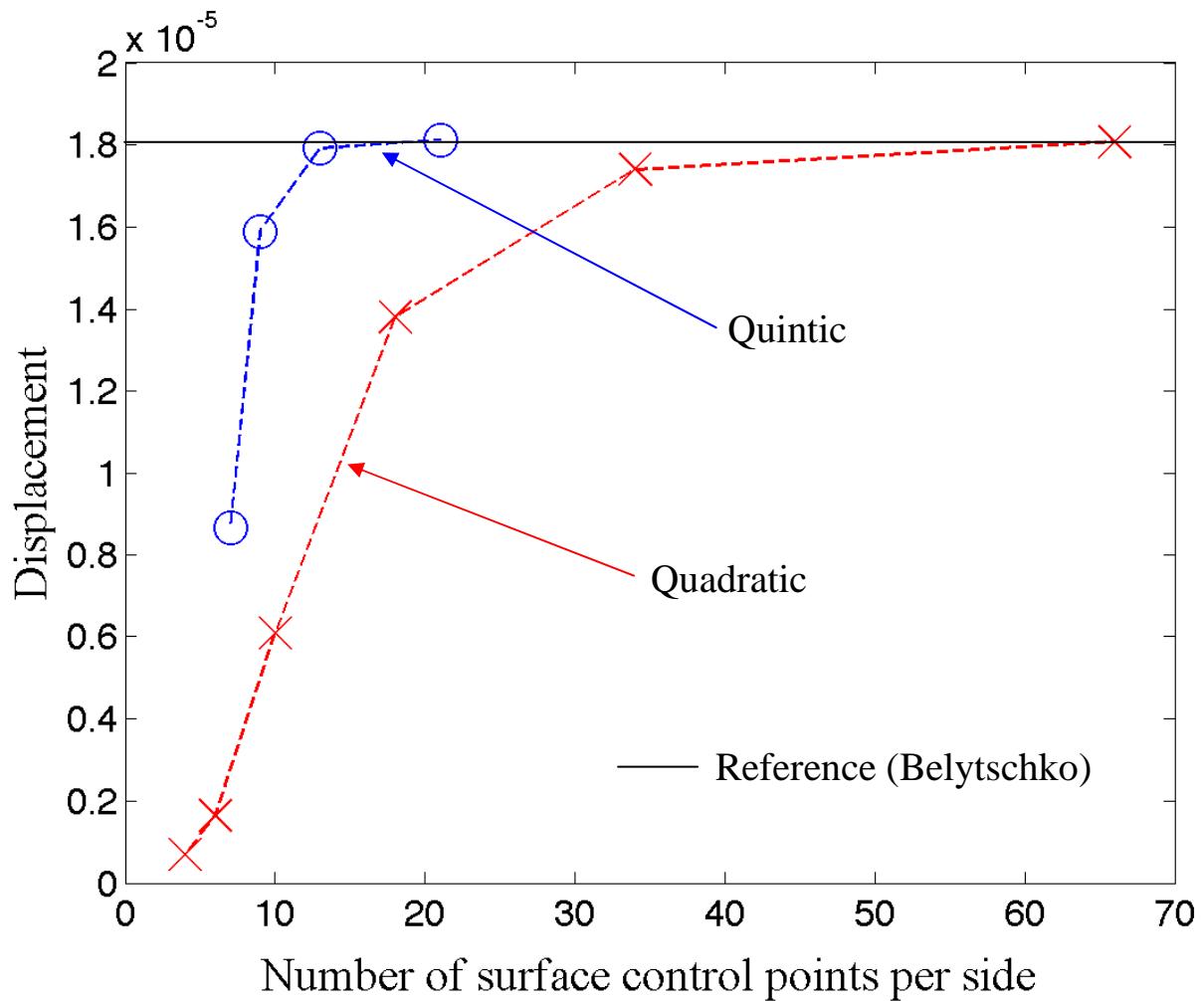


# Eigenvalue Results:

## Simply Supported Flat Plate with Quadratic NURBS



# Pinched Cylinder Displacement Convergence for NURBS Meshes



# Flat Plate Problem

- Data from Belytschko et al., *CMAME* 42 (1984) 225-251.
- Simply supported  $10 \times 10$  plate subjected to uniform pressure load of 300 psi.
- Reference solution:  $64 \times 64$  B-T elements.
- Isogeometric solutions for  $P=2,3,4$
- IFORM=3



# Flat Plate Problem

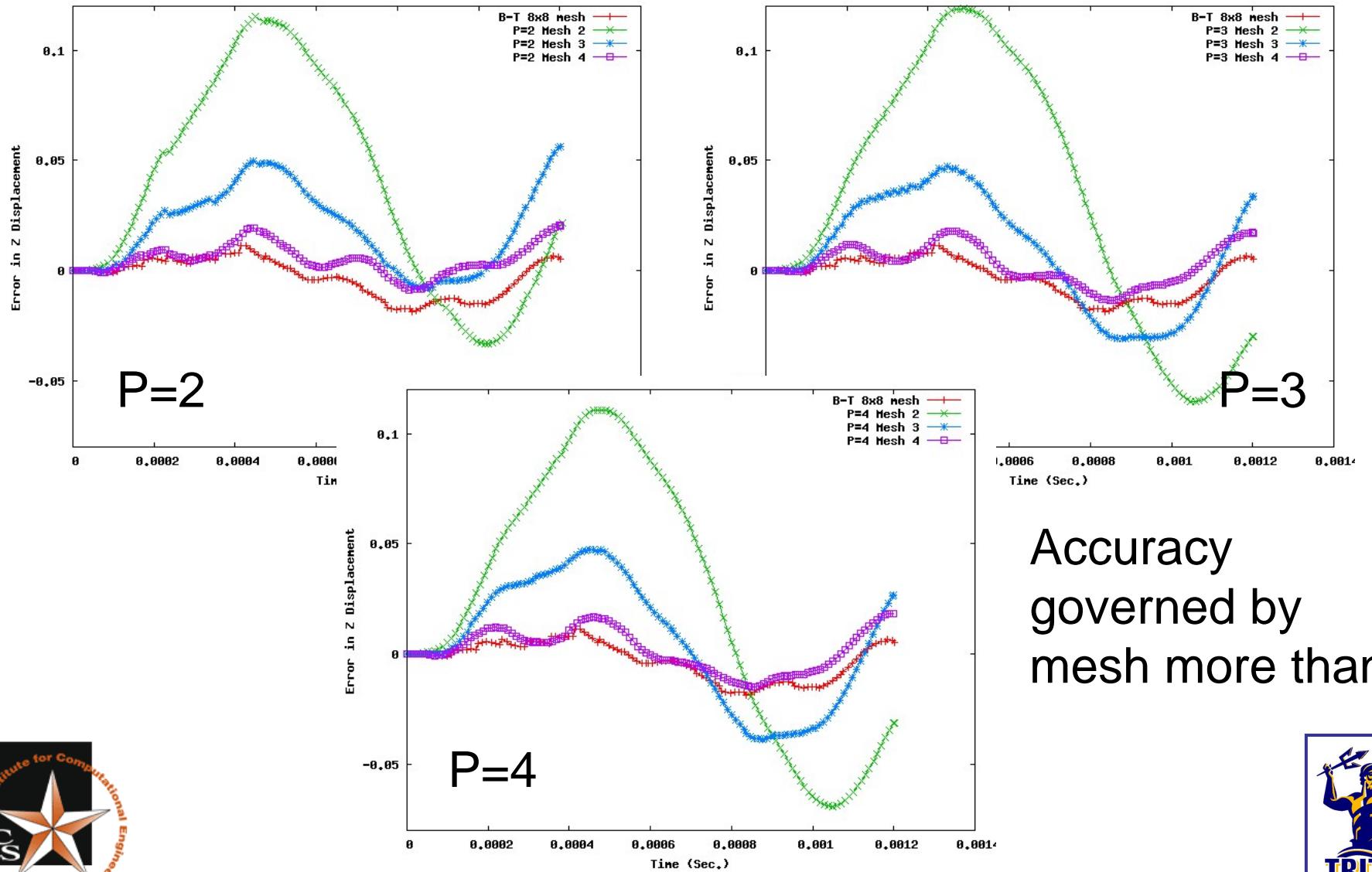
## Isogeometric Meshes

	P=2	P=3	P=4
M=2x2	16 CP	25 CP	36 CP
M=4x4	36 CP	49 CP	64 CP
M=8x8	100 CP	121 CP	144 CP

In 1-D: # Control Points (CP) = P + # Elements



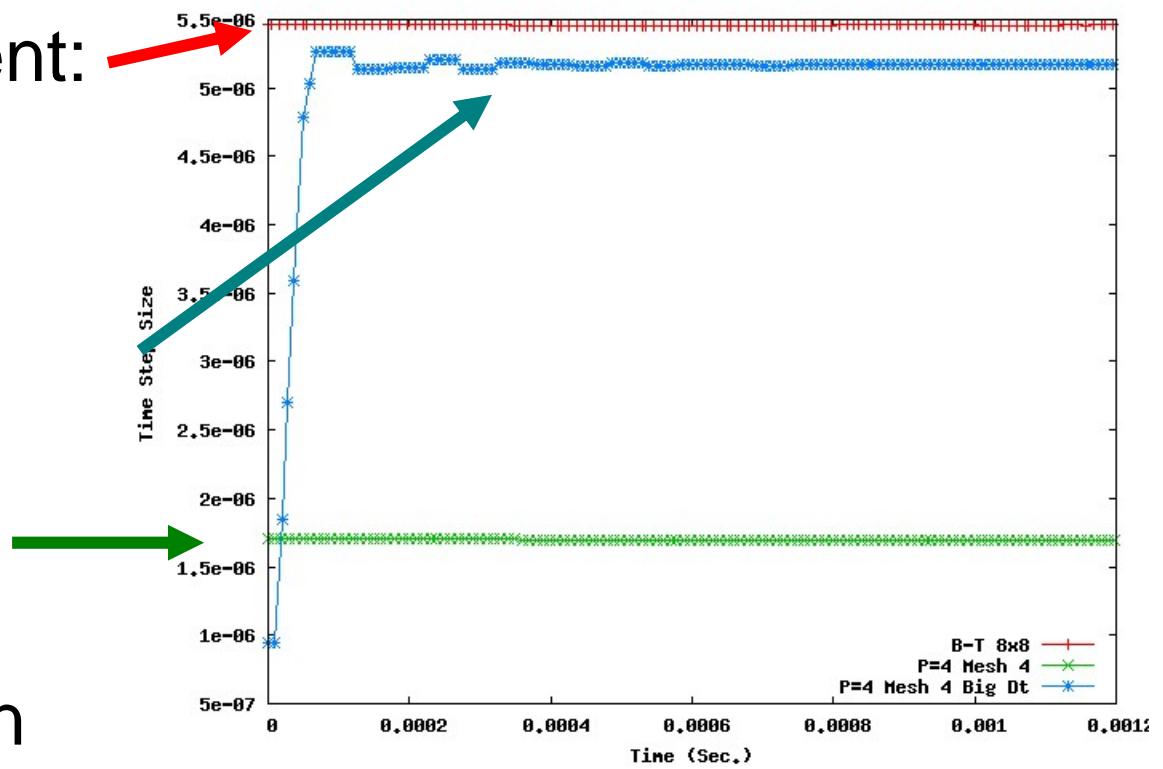
# Flat Plate Convergence Study



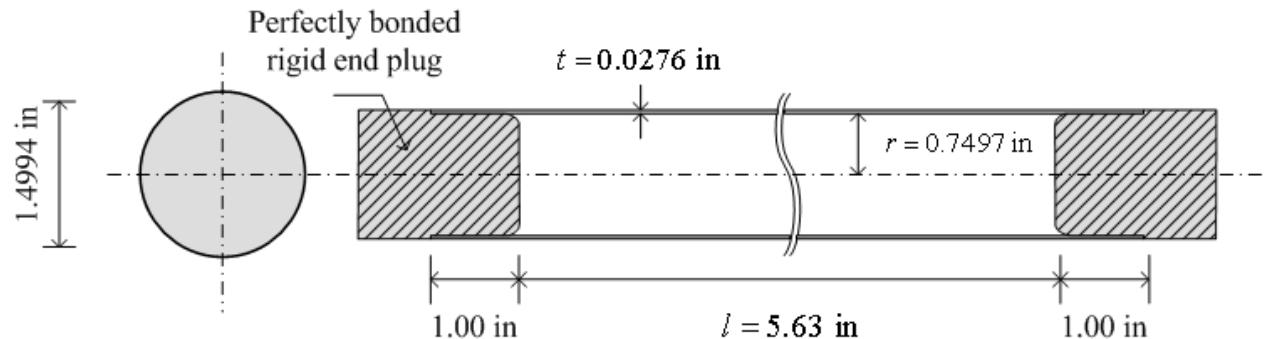
# Flat Plate Time Step Size

Preliminary Result: Higher Order NURBS Have Dt  
Almost as Large as Linear Elements

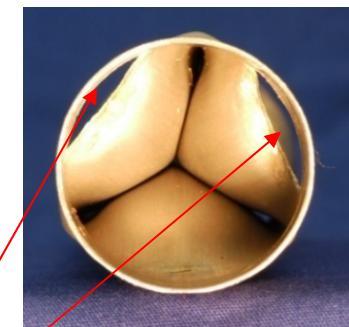
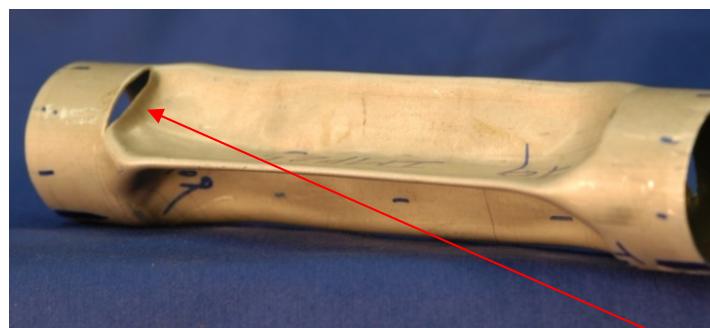
- Linear B-T element:  
 $dt=5.5 \times 10^{-6}$
- Quartic NURBS using maximum eigenvalue:  
 $dt=1/\omega=5 \times 10^{-6}$
- Quartic NURBS element based on control point spacing:  
 $dt=1.7 \times 10^{-6}$



# Texas Experiment(IMP25): Short Cylinder (Mode 3) (1)



- **Experiment:**  $p_{cr} = 410$  psi



(Fracture at the ends was observed.)

- **Analytic solution:**  $p_{cr} = 441.43$  psi

- **Computed critical pressure:**  $p_{cr} = 410$  psi

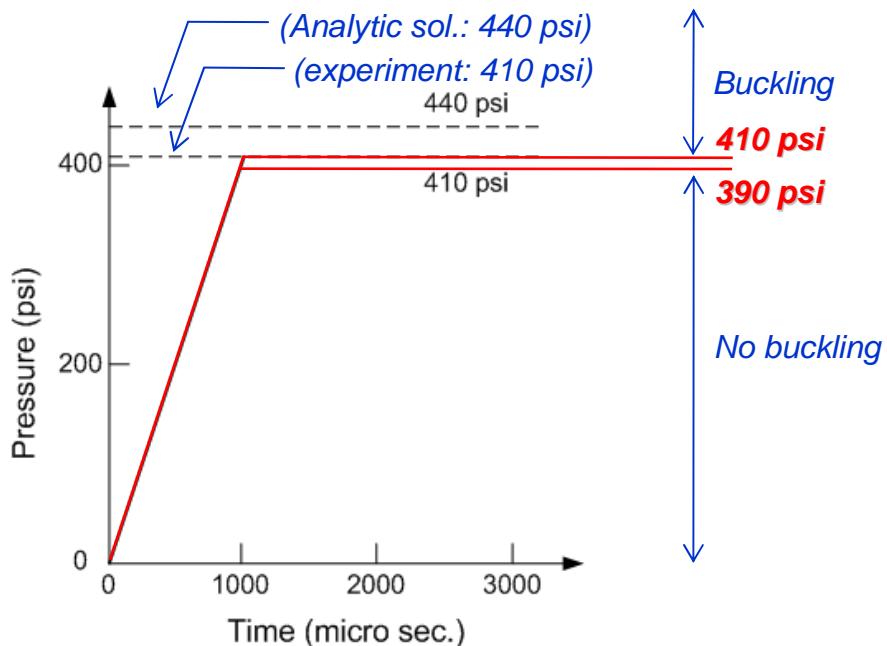


Data courtesy of Ted Belytschko  
Experimental results courtesy of Stelios Kyriakides, Univ Texas



# Texas Experiment(IMP25): Short Cylinder (Mode 3) (3)

- **Prediction of buckling pressure:**  
*(Numerical prediction)*



- **Experiment:**  $p_{cr} = 410 \text{ psi}$

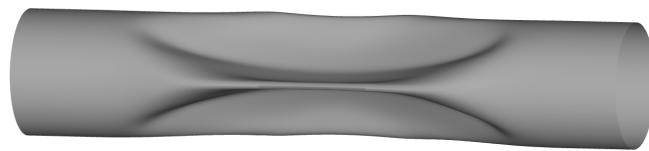
22800 Elements.  
Linear interpolation.



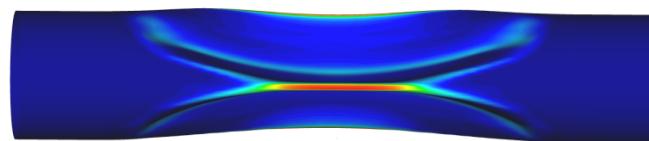
Reference solution courtesy of Ted Belytschko

$$p = 410 \text{ psi } (t = 20.0 \text{ ms})$$

Deformed shape



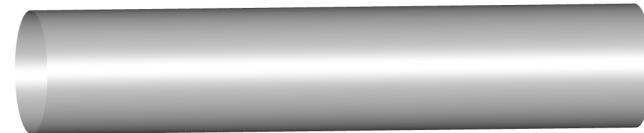
Effective plastic strain



0.17076
0.15179
0.13282
0.11384
0.094868
0.075895
0.056921
0.037947
0.018974
0

$$p = 390 \text{ psi } (t = 20.0 \text{ ms})$$

Deformed shape

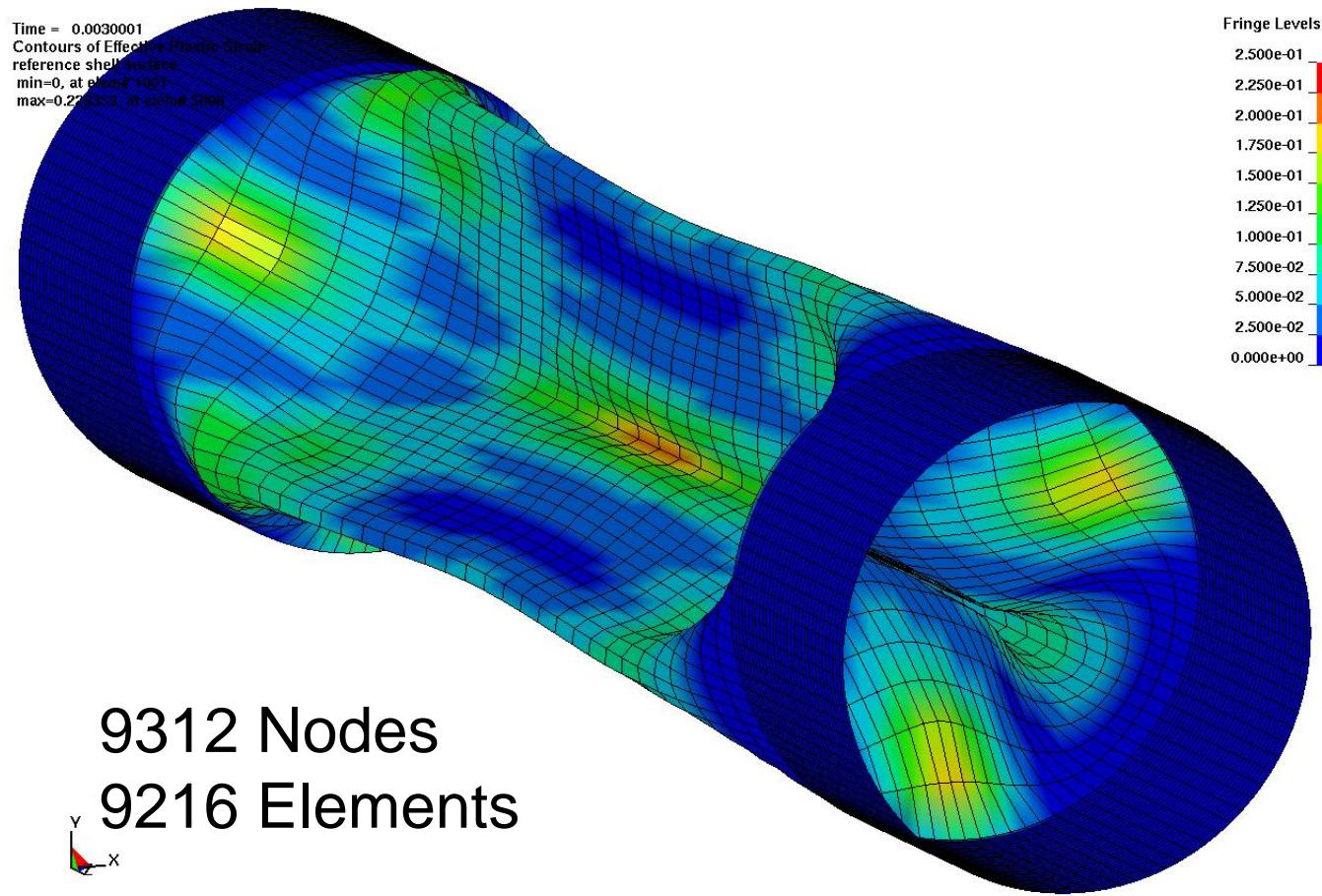


(No buckling or plastic deformation is observed.)



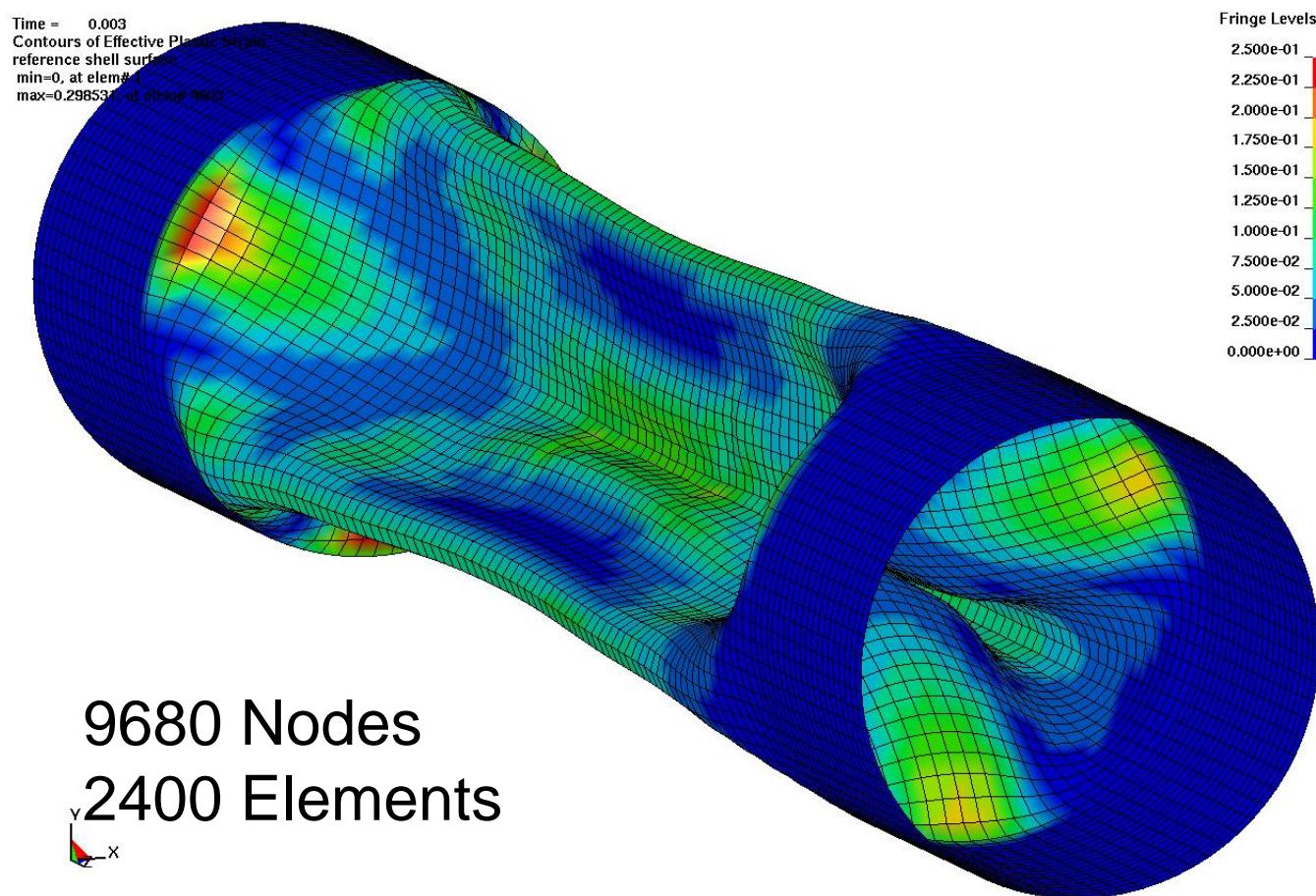
# Short Cylinder Experiment

Effective Plastic Strain at Mid-Surface  
Production Belytschko-Tsay Element  
Linear C<sup>0</sup> Basis Functions



# Short Cylinder Experiment

Effective Plastic Strain at Mid-Surface  
Quadratic C<sup>0</sup> Basis Functions; IFORM=3

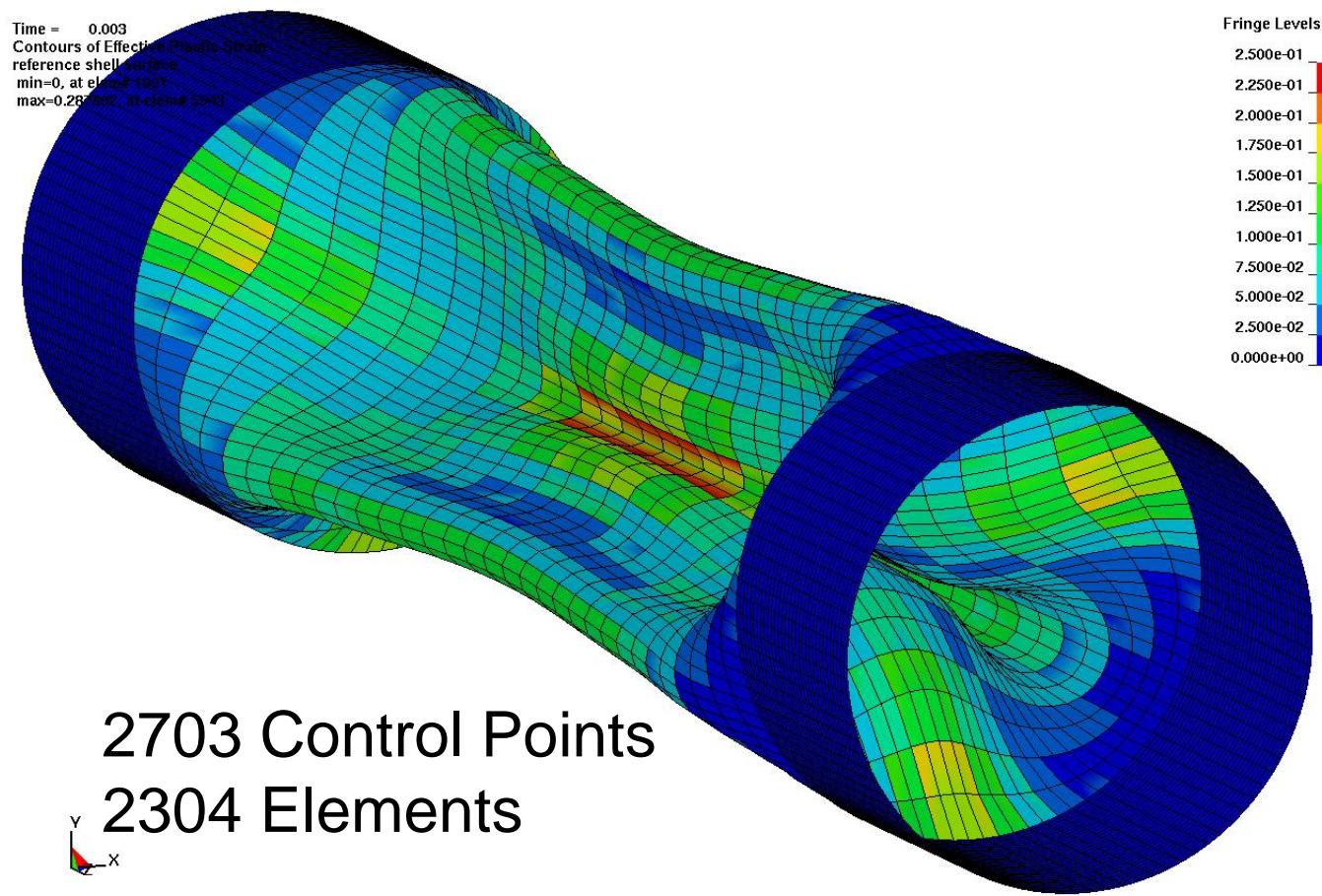


9680 Nodes  
2400 Elements



# Short Cylinder Experiment

Effective Plastic Strain at Mid-Surface  
Quadratic C<sup>1</sup> NURBS; IFORM=0



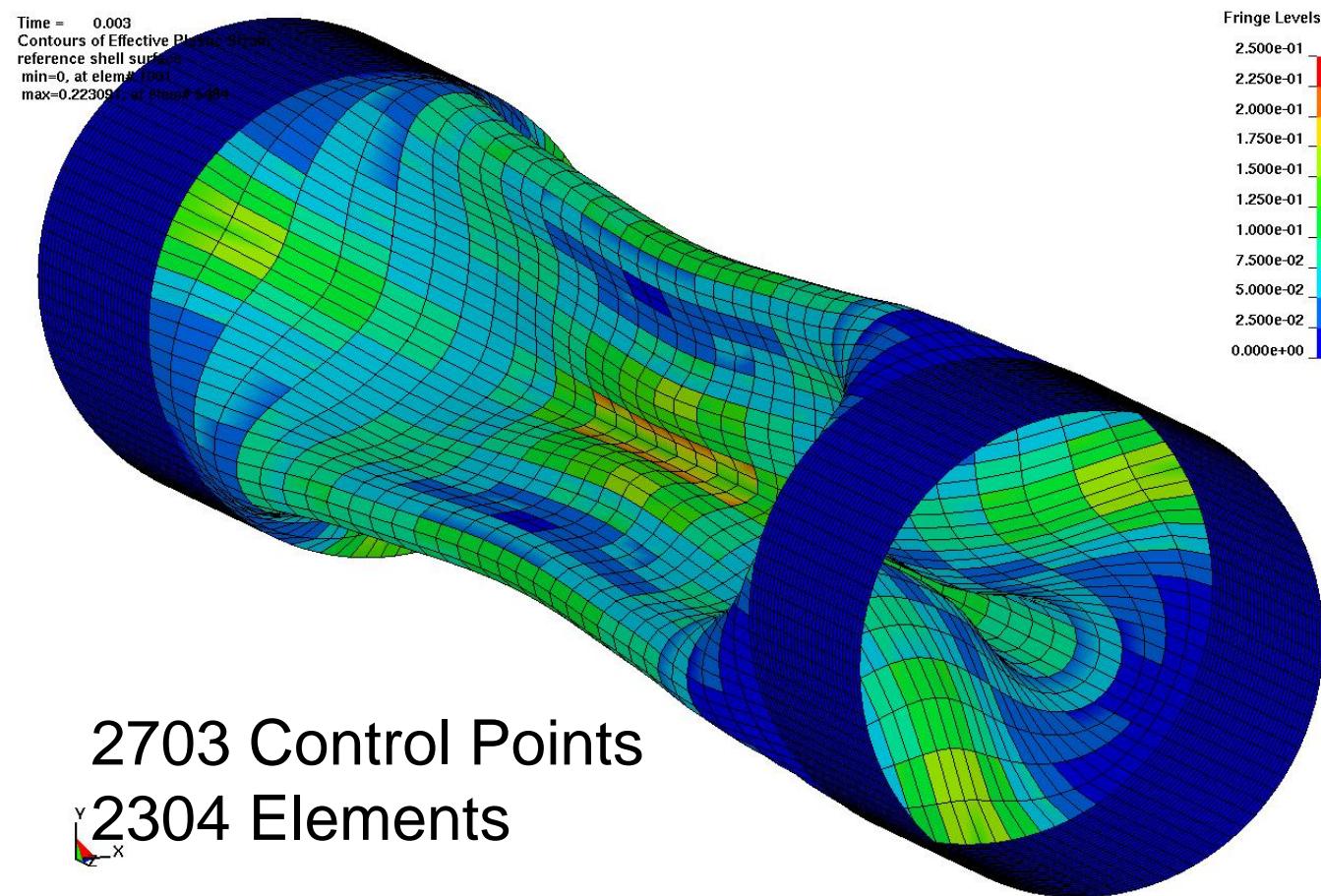
2703 Control Points  
2304 Elements



# Short Cylinder Experiment

## Effective Plastic Strain at Mid-Surface

### Quadratic C<sup>1</sup> NURBS; IFORM=3



# User-Defined Element Speed

- No performance data presented because:
  - Current implementation is for research, not production.
  - Anticipate future research to improve performance just a 4-node shells improved to current B-T speed.
- User-defined element speed is:
  - Fast enough for implicit.
  - Slow for explicit.



# Expected CPU Cost Scaling

- User-defined element overhead:
  - Generality of # nodes and # integ. points.
  - Interpolation elements for contact.
  - Interpolation nodes for force distribution.
- Belytschko-Tsay: 1 point integration & 4 nodes.
- User-defined (U-D) element: G integration points & N nodes.
- U-D/Belytschko-Tsay cost:
  - Cost ratio  $\sim (G/1)(N/4)$  ignoring matl. model.
  - Example: 9-node  $C^0$  Lagrangian cost ratio =  $81/4 \sim 20$
  - Measured:  $62/1.5 \sim 40$
- Expect production code to be at least twice as fast as user-defined element without advances in formulation.



# Current CPU Costs Due to Shell Formulation

- True curved element formulations.
- Full integration -- loss of simplifications associated with 1-point integration.
  - Integration is area of current research.
  - Early results indicate possible dramatic speed improvements.
- IFORM=2 & 3 have terms involving derivatives of the normal vector (require second derivatives of basis functions).



# Summary & Conclusions

- Ultimate goal: direct CAD to analysis.
- Isogeometric elements available in LS-DYNA via user-defined elements.
  - Explicit.
  - Implicit.
  - Eigenvalue analysis.
- Preliminary results suggest that:
  - Higher order accuracy without dt penalty of Lagrangian shape functions.
  - Excellent eigenvalue accuracy.
  - Shell elements without rotational DOF show promise.
  - Robust large deformation performance.

